COPY NO. 28

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PILOT'S

ABBREVIATED CHECKLIST

MODIFIED AIRCRAFT



7 September 1965

Changed 10 June 1966

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A

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- 3. Crossfeed & boost pumps Press on
- 4. Pump release Actuate
- 5. Tanks 1, 2, & 6 Check on
- 6. Crossfeed Press off
- 7. Fuel quantity Check
- 8. Gear warning lights Test
- 9. Ind. test Press
- 10. Headset and mask Connect (if suit not used).
- 11. Oxygen systems ON (if suit not used).
- 12. Tape and flight recorders ON

STARTING ENGINES

- T. Check with INS crew
- Fuel low pressure lights Off.
- 3. Engine instruments Check
- 4. Starter Call ready for start
- 5. Throttle IDLE at rpm rise
- 6. Fuel flow Check
- 7. Verify ignition within 15 seconds by continuous rpm and EGT increase
- 8. EGT Check for 540 C max
- 9. Starter Call off at 3300 rpm
- 10. Idle rpm Check 3550-3650 rpm
- Engine and hydraulic instruments -Normal
- 12. UHF BOTH
- 13. Other engine -Use same procedure
- 14. TEB counter Check
- INS mission only

CLEARING ENGINE

- Throttle OFF 1.
- Starter Crank 15 sec, then call OFF 2.

BEFORE TAXIING

- UHF and IFF/SIF Check 1.
- IFF As required 2.
- Generators RESET at idle rpm 3. (Check with INS crew before resetting)
- Battery BAT (within 3 sec) 4.
- Generator Out lights Check off 5.
- INS DEST/FIX VARIABLE DEST
- 6. T INS Mode - NAV (Check with INS crew prior to actuating switch.) Press STORE button and check BDHI No. 2 needle for 10° Rt. DTG 122 N Mi.
- INS Report data when slew complete
- INS DEST/FIX VARIABLE FIX and STORE-FIX REJECT light on
- INS DEST/FIX VARIABLE DEST and 10 STORE-FIX REJECT light off
- INS umbilical cord Disconnected Π (Confirmed by INS crew.)
- External power Disconnect 12.
- Forward bypass Confirm both open 13.
- HF radio ON 14.
- SAS channel switches ON DA AERO
 - SAS recycle lights Press off 16.
 - SAS light test switch Press 17.
 - Autopilot pitch and roll Engage 18.
 - Autopilot control stick disengage Press 19.

- 20. SAS channel switches OFF
- 21. Surface trim Check & set to zero
- 22. Control system Check
- 23. Packages As required
- 24. Canopy & seat pins Remove & stow
- 25. Canopy Close and lock
- 26. Canopy seal pressure ON
- 27. Rear view mirror Check
- 28. Taxi clearance Obtain
- 29. Chocks and gear pins Removed
- 30. Steering Engage and check

TAXIING

- 1. Brakes Check
- 2. Flight instruments Check
- 3. Nav eq'pt Check TACAN, ADF, INS

BEFORE TAKEOFF

- 1. Engine trim As required
- 2. SAS channels Engage
- 3. SAS lights Check off
- 4. Surface trim Check zero
- 5. Tanks 1, 2, & 6 ON
- 6. INS Check and fix as required
- 7. Compasses Check and sync FRS
- 8. Pitot heat On
- 9. Warning lights Off (except MANUAL AEROSINLET).
 - 10. Shoulder harness Lock
 - 11. BCN lights As required

- 12. Flight controls Cycle & check hydro pressure
- 13. Suit vent boost NORM
- 14. B-W ON

TAKEOFF

- 1. Brakes Hold
- 2. Elapsed time clock Start
- 3. Steering Check engaged
- 4. Throttles Advance
- 5. Brakes Release at 6000 rpm
- 6. Engine instr. Check at MILITARY
- 7. Throttles Afterburner mid-range
- 8. Throttles MAX THRUST
- 9. Engine instr. Check at MAX THRUST
- 10. Acceleration Check
- 11. Rotation Begin at computed KIAS

AFTER TAKEOFF

- l. Gear UP
- 2. Throttles Climb power
- 3. Engine instr. Check
- 4. Surface limiter In
- 5. Fuel derich ARM

NORMAL CLIMB

- 1. Airspeed Establish climb schedule
- 2. Altimeter Set 29.92 at FL 180

N-8

- 4. Canopy Open
- 5. Igniter purge DUMP
- 6. Recorders OFF
- 7. Appropriate electrical switches Off
- 8. Inverters OFF
- 9, Battery OFF
- 10. Generators TRIP
- 11. Throttles OFF
 - 12. Seat and canopy pins Installed

TAKEOFF AND LANDING DATA CARD

(Refer to front flap of checklist)



DOUBLE ENGINE FAILURE IMMEDIATELY AFTER TAKEOFF

- 1. IF GEAR IS DOWN AND CONDITIONS PERMIT LAND STRAIGHT AHEAD.
- 2. IF GEAR RETRACTION HAS BEEN INITIATED OR CONDITIONS DICTATE EJECT

AFTERBURNER NOZZLE FAILURE

Nozzle Failed Open Immediately After Takeoff

- 1. Throttle AB range
- 2. Monitor rpm and EGT
- 3. Land as soon as possible

Nozzle Failed During Cruise

- Throttle MILITARY or below
- 2. Monitor rpm and EGT
- 3. Land as soon as possible

AFTERBURNER FLAMEOUT

- 1. Throttle MILITARY
- 2. Throttle AB midrange (note TEB)
- 3. Nozzle position Check

If start not successful:

Throttle - MILITARY

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INLET DUCT UNSTART

- 1. SIMULTANEOUSLY REDUCE ANGLE OF ATTACK, BOTH RESTARTS ON
- 2. BOTH THROTTLES MILITARY
- 3. MAINTAIN ATTITUDE CONTROL OPTIMIZE PITCH AND ROLL
- 4. AIRSPEED ADJUST TOWARD 350 KEAS & DO NOT EXCEED MACH 3.1

If roughness not clear in 10 seconds:

5. AFT BYPASS - OPEN

When roughness clears:

- 6. Aft bypass Normal schedule
- 7. Fwd bypass Both Open
- 8. Restart Both OFF

After inlet starts:

- 9. Fuel derich Recycle below 790°C EGT if actuated.
- 10. Throttles As required
- 11. Fwd bypass Both AUTO

If unstarts repeat or inlet doesn't clear:

- 12. Engine, inlet instr, hyd press Check
- 13. Repeat procedure

If unstarts persist:

14. Attempt restart and operation using manual inlet operating schedule

E-4

- . - (MOD.)

MANUAL INLET OPERATING SCHEDULE

Manual Spike Schedule

Accelerating - Lag Mach by 0.1

Cruising - Match Mach number

Decelerating - Lead Mach by 0.1

Manual Bypass Schedule

Mandatory with manual spike.
Optional with auto spike and other inlet operating normally.

Condition	Mach	Fwd Bypass	Aft Bypass
Accel. & cruise	Above	Pos. 7	Pos. B
Accel. & cruise	Above 2.8	Pos. 8	CLOSED
Decel	ALL	OPEN	CLOSED

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JADA MEROS

AIR INLET CONTROL FAILURE

SPIKE NOT FWD light not on with SPIKE FWD selected

 Check L or R hydro press normal & MANUAL INLET light on

If hydraulic failure has occurred and flight and mission conditions dictate:

2. Emergency spike switch - FWD.

Spike not scheduling or inlet spike unstable

- 1. Spike position ind. Check
- Spike Cycle FWD then return to AUTO

If condition continues:

- 3. Forward bypass Manual schedule
- 4. Spike Manual schedule

As higher Mach number is reached:

5. Spike and forward bypass - AUTO.

If condition recurs or continues:

6. Operate per spike and bypass manual schedule

E-6

HALL OF FAM

6-10-66

-. - (MOD.)

ELECTRICAL POWER SYSTEM FAILURE

SINGLE AC GENERATOR FAILURE

- 1. Generator RESET
- If light remains on:
 - 2. Generator TRIP.
 - 3. Land as soon as possible
- If flight continued:
 - 4. Affected generator TRIP
- If EWS is operating:
 - 5. TACAN OFF

DOUBLE AC GENERATOR FAILURE

- l. Battery BAT
- 2. Generators RESET
- 3. If only one generator resets Land
- 4. If neither generator resets Conserve batteries and land as soon as possible

INVERTER FAILURE

- 1. Failed inverter EMER
- 2. Illuminated SAS lights Press

HYDRAULIC POWER SYSTEM FAILURE

L-HYDRAULIC SYSTEM FAILURE

- a. Be prepared to use L EMER SPIKE AFROFWD SWITCH
- b. Emergency gear extension required
- c. Use alternate brakes & NWS

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- . - (MOD.)

R-HYDRAULIC SYSTEM FAILURE

a. Be prepared to use R EMER SPIKE FWD SWITCH

FLIGHT CONTROL SYSTEM FAILURE

FLIGHT CONTROL SYSTEM EMERGENCY OPERATION

If control difficulties are encountered:

1. Check A and B hydraulic pressures

If neither A or B hydraulic system has failed:

- Disengage autopilot, check control
- 3. Check SAS warning lights. If SAS failure has occurred, see SAS Emergency Operation

A OR B HYDRAULIC SYSTEM FAILURE

- Reduce KEAS to less than 350
- Affected SAS yaw and pitch channels -OFF
- 3. SAS roll channels Both Off
- Operative roll channel ON
- 5. Hyd. Res. oil Operative system A or B.

A AND B HYDRAULIC SYSTEMS BOTH FAILED

1. EJECT

E-20

HALL OF FAME

SAS EMERGENCY OPERATION

- Check A and B hydraulic pressures -Normal
- 2. Check INVERTER OUT warning lights not illuminated
- Proceed to appropriate Roll Axis or Pitch or Yaw Axis Failure procedure

ROLL AXIS FAILURE

- A or B channels OFF then ON
 If light extinguished a transient probably existed and both roll channels are engaged:
 - 2. After light extinguishes establish momentary roll transients

If light does not extinguish or reilluminates:

- 3. A and B channels OFF
- 4. A channel engage ON

If no improvement is noted:

- 5. A channel engage OFF
- 6. B channel engage ON

PITCH OR YAW AXIS FIRST FAILURE

- 1. Refer to SAS Failure Warning Lights chart
- 2. Recycle indicator light Press and release

If light extinguishes:

3. Duplicate maneuver which caused light to illuminate

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If light remains on or reilluminates:

- 4. Faulty channel switch OFF
- 5. Decelerate to pitch or yaw axis second failure limit speed if conditions permit.
- 6. Evaluate damping on remaining channel

PITCH AXIS SECOND FAILURE

- 1. Airspeed 350 KEAS maximum
- 2. Remain supersonic and use logic override procedures if appropriate

If conditions permit:

- 3. Descend
- Forward fuel transfer ON Maintain 4000 pounds in tank 1
- 5. Airspeed Maintain Mach 1.3 min. until fuel forward transfer is complete
- 6. Airspeed Slow to best subsonic cruise speed and altitude
- 7. Refer to BUPD emergency procedure If BUPD cannot be used:
 - 8. Use caution to avoid abrupt maneuvers during landing approach

YAW AXIS SECOND FAILURE

- l. Max. airspeeds Mach 2.5
- 2. Remain supersonic and use logic override procedure if appropriate

Or if conditions permit:

- 3. Descend
- 4. Airspeed Slow to best subsonic speed and altitude for duration of flight

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LANDING GEAR SYSTEM EMERGENCY OPERATION

RETRACTION

- 1. Ground retract button Press and hold
- 2. Gear lever UP

EXTENSION

- 1. Gear lever DOWN
- 2. Emergency gear handle PULL
- 3. Verify gear down and locked

If landing gear remains retracted:

- 4. Gear CONT c/b PULL
- 5. Repeat steps 2 and 3

WHEEL BRAKE SYSTEM FAILURE

BRAKE SYSTEM EMERGENCY OPERATION

1. Brake switch - ALT STEER & BRAKES

AIR DATA COMPUTER FAILURE

- Check TDI, airspeed and altimeter
 If cross check shows TDI to be inaccurate:
 - 2. Revert to pitot-static instruments
 - 3. Pull MACH TRIM c/b
 - 4. Autopilot OFF

PITOT-STATIC SYSTEM FAILURE

1. Attempt operation on alternate source

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- 2. Maintain control by use of attitude and power indicating instruments
- 3. Request escort aircraft

AIR CONDITIONING & PRESSURIZATION FAILURE

LEFT ENGINE INOPERATIVE

Cockpit system - CROSSOVER

COCKPIT AND SUIT OVERTEMPERATURE

- l. Defog OFF
- 2. Cockpit temp ind. Check

If temp ind is too high:

- 3. Cockpit auto temp Rotate to COLD If cockpit temp remains high:
- 4. Cockpit temp switch Hold in COLD If no temperature decrease:
 - 5. Cockpit system CROSSOVER
 - 6. Q-Bay system Check ON
- If suit temperature cannot be controlled:
 - 7. Suit flow valves OFF
 - 8. Reduce altitude and speed

Q-BAY OVERTEMPERATURE

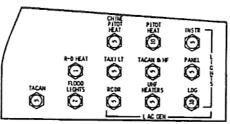
- 1. Q-Bay auto temp Rotate to COLD If not effective:
 - 2. Q-bay temp control Hold in COLD

COCKPIT DEPRESSURIZATION

If suit inflates:

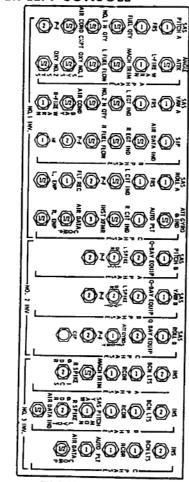
1. Cockpit altitude - Check.

E-2.8

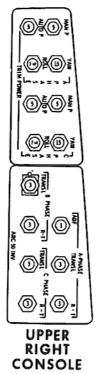


UPPER LEFT CONSOLE

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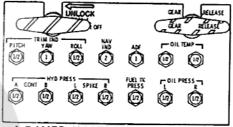
RIGHT SIDE-A.C.



RIGHT CONSOLE

LEFT CONSOLE





LOWER INSTRUMENT PANEL

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